

### Mechanical Performance and Surface Control of Additively Manufactured Aluminum Hardware

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2023 Contamination, Coatings, Materials, and Planetary Protection (CCMPP) Workshop September 12 – 14, 2023 Greenbelt, Maryland

# Additive manufacturing (AM) is an attractive process for complex geometries and can enable novel radiofrequency (RF) applications.



The internal stepped geometry of a septum polarizer is challenging to fabricate using traditional machining

References:

A. Sharma and C. Carpenter, 2023 USNC-URSI NRSM (2023)

A. Sharma, 2021 IEEE APS/URSI (2021)



Machined and Dip Brazed Aluminum

AM Aluminum

## Use of AM for spaceflight must ensure 1) mechanical properties are sound and 2) the surface is sufficiently controlled for contamination.



Use of lot-specific witness samples helps validate mechanical properties



Vibration testing of prototype and flight builds serves as a proof test for mechanical integrity



Development of a plating process addresses particulate control concerns

Hardware is built inside a bed of powder in which the height of the bed is controlled on a moving platform.





To build a layer of the part, the build platform lowers and the powder platform is raised.





Part Powder **Recoater Bar Powder Platform Build Platform** 

The recoater bar is swept across the build platform surface to fill the bed with powder.

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A laser is rastered across the surface to weld localized areas of the bed, creating a continuous part.





The process is completed layer by layer until the desired part is created.





# A6061-RAM2 powder was selected for development of a low gain antenna for its improved surface roughness over AISi10Mg.



Reference: A. Sharma, C. Carpenter and J. Dennison, 2020 IEEE AP-S/URSI (2020)



## Laser-powder bed fusion (L-PBF) can produce low porosity hardware similar to wrought when the AM process is optimized.





# A design of experiment (DOE) was used to determine optimized build parameters for the laser-powder bed fusion (L-PBF) process.

#### YS (MPa)



Summary Stat	Good	
Mean	245.56579	
Std Dev	21.791947	Shapiro
Std Err Mean	2.4997077	
Upper 95% Mean	250.54546	Anders
Lower 95% Mean	240.58612	
Ν	76	
Skewness	-0.796377	
Minimum	187	
Maximum	277	
Range	90	

Goodness-of-Fit Test					
		W	Pr	ob <w< th=""><th></th></w<>	
Shapiro-Wilk	0.93	349365 <mark>0</mark> .		.0007*	
			A2	Prob	> A2
Anderson-Dar	rling	1.4406	433	0.0	010*

- ASTM E8 tensile test coupons, N = 76
- Includes all of the laser parameter sets from the DOE #2 parameter optimization study
- Data is skewed to lower YS values due to high porosity material that also had low strength (some higher porosity material had high strength)
- Goodness-of-Fit Test shows that the data is not normally distributed, although visually the shape isn't too bad

## A design of experiment (DOE) was used to determine optimized build parameters for the laser-powder bed fusion (L-PBF) process.

Property	DOE All Results	DOE Low Porosity
Number of Samples	76	45
UTS (MPa)	274 (mean) 244 (T99)	277 (mean)
YS (MPa)	246 (mean) 203 (T99)	255 (mean)
Elongation (%)	7.6 (mean) 4.7 (min)	_



## Witness tensile coupons build alongside flight antennas exhibited excellent mechanical properties.

Property	DOE All Results	DOE Low Porosity	Witness Coupons	Elementum 3D Datasheet	Wrought 6061-T6
Number of Samples	76	45	6	_	_
UTS (MPa)	274 (mean) 244 (T99)	277 (mean)	317 (mean)	331	310
YS (MPa)	246 (mean) 203 (T99)	255 (mean)	291 (mean)	297	276
Elongation (%)	7.6 (mean) 4.7 (min)	_	8.9	11-14	12-17

CONCLUSION: Mechanical properties were satisfactory for our spaceflight application.



Plating was used to address contamination concerns—corrosion protection, particulate encapsulation, and improved cleaning.









2023 CCMPP

### A modified zincate process was utilized to achieve an electroless nickel seed layer, followed by electrodeposited gold.



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## Coverage of the plating over the complex geometry was excellent, with few areas of thin or missing gold.



## Both a plated prototype and plated witness coupons were subject to thermal cycling, both in air and in high vacuum.





Thermal cycling was completed in a two-chamber system, in which samples on an elevator moved between a hot and cold chamber rapidly with 15 minutes dwell at temperature. Temperature change is near 36°C/min at maximum rate.

## The plating was fully adherent and unaffected by thermal cycling, as determined by visual inspection and an ASTM D3359 tape pull.



**Before Thermal Cycling** 



After Thermal Cycling



## The finished antenna passed vibration testing and met all RF requirements.



CONCLUSION: AM is a promising route for future RF hardware.





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